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## Solutions of 2D and 3D Stokes laws using multiquadrics method

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### Abstract

In this paper, velocity–vorticity formulation and the multiquadrics method (MQ) with iterative scheme are used to solve two (2D) and three-dimensional (3D) steady-state incompressible Stokes cavity flows. The method involves solving of Laplace type vorticity equations and Poisson type velocity equations. The solenoidal velocity and vorticity components are obtained by iterative procedures through coupling of velocity and vorticity fields. Both the Poisson type velocity equations and the Laplace type vorticity equations are solved using the MQ, which renders a meshless (or meshfree) solution. Here, the results of 2D Stokes flow problems in a typical square cavity and a circular cavity are presented and compared with other model results. Besides utilizing the MQ to solve the 3D Stokes cubic cavity flow problem, we are also obtaining promising results for the accuracy of the velocity and vorticity. The MQ model has been found to be very simple and powerful for analyzing the 2D and 3D internal Stokes flow problems.

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### 1. Introduction

Stokes flow is a classical problem in fluid dynamics. After many years of development, three formulations for numerical analysis of the Stokes flow are most known: vorticity–stream vector function or vorticity–vector potential approach, primitive variable (velocity–pressure) approach, and vorticity–velocity approach. Using the vorticity–velocity approach, we can transfer the governing equations of primitive variable Stokes equations into a system of Laplace and Poisson equations for the components of vorticity and velocity fields. By doing so, the direct computation of pressure that is rather difficult to deal with is avoided.

Among various methods for computational fluid dynamics (CFD), the finite difference method (FDM), the finite element method (FEM) and the boundary element method (BEM) are the three most famous numerical schemes adapted for mesh generation. Although the field of computational science was developed widely and quickly, there still exist some unresolved problems for

these numerical methods. The following drawbacks of the conventional numerical schemes with mesh are observed: the awkward treatment of irregular boundary in FDM; the storage of huge data in FEM; the difficulty of the singularities and fundamental solutions in BEM; etc. In this study, an innovative scheme—the multiquadrics method (MQ) or the so-called Kansa's method [1,2] is used to deal with the above-mentioned shortcomings and also to provide the meshfree scheme.

MQ scheme is a truly scattered, grid free (or meshless) scheme for representing surfaces and bodies in an arbitrary number of dimensions. The radial basis functions (RBF) depend only upon distances between pairs of points. So, it is very powerful for dealing with irregular domain problems. MQ scheme is a very promising method not only for very accurate interpolation of functions, but also for their partial derivatives, divergences, curls, gradients, and integrals. So, error analysis and solving of differential and integral equations become much easier than with the use of the conventional numerical methods. By assuming the suitable interpolation functions—the radial basis functions, there will be no fundamental solutions or singular integration problems involved, such as in the BEM. Madych and Nelson [3] also have proved the theoretical justification for MQ's performance and gave us confidence to justify our numerical

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results. Recently, Cheng et al. [4] also have demonstrated the exponential convergence for the MQ collocation method, if the free shape parameter is chosen properly.

MQ was used to simulate the shallow water problems by Hon et al. [5]. They used the partial derivatives of radial basis functions and scattered collocation points to solve the shallow water equation problems in irregular topographic water bodies. As for the size of the region, the resolving way of collocation, and the rank of matrix, Kansa and Hon [6] discussed the ill conditioning of the problem applied to elliptic partial differential equations. They explore several techniques to improve the condition of the coefficient matrix to get rid of the ill conditioning of matrix inversion and achieve the more accurate results. The shape parameter in the radial basis functions is another issue in hot debate nowadays. Kansa and Carlson [7], for example, investigate the usage of variable shape parameters to improve the accuracy of multiquadric interpolation. An adaptive collocation method to solve the Burger's equation that is a nonlinear partial differential equation is proposed by Hon and Mao [8]. More descriptions of the MQ scheme to solve various linear and nonlinear partial differential equations can be referred to Fedoseyev et al. [9].

Sometimes, in order to increase the accuracy of the result, more points may be required to represent the configuration and characteristics of a region. Therefore, solving large radial basis function interpolation problems encounter some difficulty in terms of computational cost and magnitude of storage and operation. Consequently, Beatson et al. [10] propose a way to resolve this problem. They adopt suitable approximate basis function with precondition and employ the GMRES iterative method to resolve these problems.

For large matrix vector problem, Beatson and Newsam [11–13] proposed to use the domain decomposition method to divide the domain into a few subdomains. The techniques are based on the hierarchical and multipole expansion for the calculation of many-body potentials. This method provides the feature for fast, storage-efficient computation of the matrix vector product.

Mei-Duy and Tran-Cong [14] used the MQ RBF to solve the 2D Navier–Stokes equations including planar Poiseuille, driven cavity and natural convection flows. They claimed that employment of the indirect radial basis function networks method (IRBFN) [15] would render better results with relatively low collocation densities. However, since the streamfunction and vorticity formulation are used, only 2D flow problems are solved. Extension to 3D problems would be very difficult.

In this paper, we combine the vorticity–velocity formulation with the MQ scheme to solve steady-state, both 2D and 3D Stokes flow problems. Three numerical examples, a square cavity problem and a circular cavity problem for 2D and a cubic cavity flow problem for 3D are employed to check the validity and efficiency of the MQ scheme. As far as the more complex Navier–Stokes flow

problems are concerned, this paper only provides a preliminary work. MQ scheme really offers an efficient and powerful as well as meshless numerical method to the realm of CFD and deserves further research and application.

## 2. Governing equations

The governing equations of Stokes flows for the velocity–vorticity formulation in steady-state can be derived from the Navier–Stokes equations by neglecting the inertia and are given as follows

$$\nabla^2 \vec{\omega} = \vec{0} \quad (1)$$

$$\nabla^2 \vec{u} = -\nabla \times \vec{\omega} \quad (2)$$

Here  $\vec{u}$  is the velocity vector, and  $\vec{\omega}$  is the vorticity vector. The vorticity vector  $\vec{\omega}$  is defined by the following

$$\vec{\omega} = \nabla \times \vec{u} \quad (3)$$

In two-dimensions, if  $(u, v)$  are the horizontal velocity vectors and  $\omega \vec{k}$  is the associated vertical vorticity vector, then the vorticity transport Eq. (1) can be expressed as

$$\nabla^2 \omega = 0 \quad (4)$$

Eq. (2) that governs the  $u, v$ -velocity becomes as follows

$$\nabla^2 u = -\frac{\partial \omega}{\partial y} \quad (5)$$

$$\nabla^2 v = \frac{\partial \omega}{\partial x} \quad (6)$$

In steady-state conditions, solution of Eq. (4), in conjunction with the Poisson Eqs. (5) and (6), gives the velocity and vorticity distributions all over the domain.

In three-dimensions, if  $\vec{u} = (u, v, w)$  are the velocity vectors and  $\vec{\omega} = (\xi, \eta, \zeta)$  are the associated vorticity vectors, then the individual vorticity transport Eq. (1) can be described as

$$\nabla^2 \xi = 0 \quad (7)$$

$$\nabla^2 \eta = 0 \quad (8)$$

$$\nabla^2 \zeta = 0 \quad (9)$$

Similarly, Eq. (2) can be deduced for 3D velocity and illustrated by the following

$$\nabla^2 u = -\left(\frac{\partial \zeta}{\partial y} - \frac{\partial \eta}{\partial z}\right) \quad (10)$$

$$\nabla^2 v = -\left(\frac{\partial \xi}{\partial z} - \frac{\partial \zeta}{\partial x}\right) \quad (11)$$

$$\nabla^2 w = -\left(\frac{\partial \eta}{\partial x} - \frac{\partial \xi}{\partial y}\right) \quad (12)$$

Similarly, being in steady-state conditions, solution of Eqs. (7)–(9), in conjunction with the Poisson Eqs. (10)–(12),

gives the velocity and vorticity distributions all over the domain in 3D Stokes flow problems.

A solution is sought in the domain  $\Omega$  that satisfies the boundary conditions,

$$\vec{u} = \vec{u}_\Gamma \tag{13}$$

and

$$\vec{\omega} = (\nabla \times \vec{u})|_\Gamma \tag{14}$$

on the boundary.

### 3. Numerical formulation

#### 3.1. Two-dimensional numerical formulation

Consider the 2D Poisson's equation

$$\nabla^2 \Phi = f(x, y) \quad (x, y) \in \text{domain} \tag{15}$$

$$\Phi = g(x, y) \quad (x, y) \in \text{boundary} \tag{16}$$

Let

$$\Phi = \sum_{i=1}^m \alpha_i r_i \sqrt{r_i^2 + c^2} \tag{17}$$

Where  $r_i$  is the Euclidean distance between the field point  $(x, y)$  and collocation point  $(x_i, y_i)$ , and  $c$  is a free shape parameter to be determined later. In general,  $0.1 \leq c \leq 1.0$  is recommended in our study. The determination of  $c$  is still an attractive research topic.  $\alpha_i$ 's are the coefficients to be determined later by the point collocation method.  $f(x, y)$  and  $g(x, y)$  are given functions. In this study,  $\Phi = u$ , or  $v$ , or  $\omega$ , and  $f(x, y) = -(\partial\omega/\partial y)$ , or  $(\partial\omega/\partial x)$ , or 0. We choose  $m$  collocation points  $\{(x_i, y_i)\}_1^m$  and use Eq. (5) for the  $u$ -velocity. The other unknowns,  $v$  and  $\omega$  can be obtained in a similar fashion.

Let

$$u = \sum_{i=1}^m \alpha_i^u r_i \sqrt{r_i^2 + c^2} \tag{18}$$

$$\omega = \sum_{i=1}^m \alpha_i^\omega r_i \sqrt{r_i^2 + c^2} \tag{19}$$

$$\frac{\partial^2}{\partial x^2} \sqrt{(x - x_i)^2 + (y - y_i)^2 + c^2} = \frac{(y - y_i)^2 + c^2}{(\sqrt{(x - x_i)^2 + (y - y_i)^2 + c^2})^3} \tag{20}$$

Similarly,

$$\frac{\partial^2}{\partial y^2} \sqrt{(x - x_i)^2 + (y - y_i)^2 + c^2} = \frac{(x - x_i)^2 + c^2}{(\sqrt{(x - x_i)^2 + (y - y_i)^2 + c^2})^3} \tag{21}$$

For the vorticity derivative of  $\omega$ ,

$$\begin{aligned} \frac{\partial}{\partial y} \sum_{i=1}^m \alpha_i^\omega \sqrt{(x - x_i)^2 + (y - y_i)^2 + c^2} \\ = \sum_{i=1}^m \alpha_i^\omega \frac{(y - y_i)}{\sqrt{(x - x_i)^2 + (y - y_i)^2 + c^2}} \end{aligned} \tag{22}$$

Substituting Eqs. (20)–(22) into Eq. (5), we obtain

$$\begin{aligned} \sum_{i=1}^m \alpha_i^u \left( \frac{(y - y_i)^2 + c^2}{(\sqrt{(x - x_i)^2 + (y - y_i)^2 + c^2})^3} \right) \\ + \sum_{i=1}^m \alpha_i^u \frac{(x - x_i)^2 + c^2}{(\sqrt{(x - x_i)^2 + (y - y_i)^2 + c^2})^3} \\ = \sum_{i=1}^m \alpha_i^\omega \frac{-(y - y_i)}{\sqrt{(x - x_i)^2 + (y - y_i)^2 + c^2}} \end{aligned}$$

Also

$$\begin{aligned} \sum_{i=1}^m \alpha_i^u \left( \frac{(x - x_i)^2 + (y - y_i)^2 + 2c^2}{(\sqrt{(x - x_i)^2 + (y - y_i)^2 + c^2})^3} \right) \\ = \sum_{i=1}^m \alpha_i^\omega \frac{-(y - y_i)}{\sqrt{(x - x_i)^2 + (y - y_i)^2 + c^2}} \end{aligned}$$

By collocation method, we obtain  $n$  equations from the interior points and  $m - n$  boundary points.

$$\begin{aligned} \sum_{i=1}^m \alpha_i^u \left( \frac{(x_j - x_i)^2 + (y_j - y_i)^2 + 2c^2}{(\sqrt{(x_j - x_i)^2 + (y_j - y_i)^2 + c^2})^3} \right) \\ = \sum_{i=1}^m \alpha_i^\omega \frac{-(y_j - y_i)}{\sqrt{(x_j - x_i)^2 + (y_j - y_i)^2 + c^2}} \end{aligned} \tag{23}$$

$$j = 1, \dots, n$$

We choose  $(m - n)$  boundary points  $\{(x_i, y_i)\}_{n+1}^m$ . From Eq. (16), we obtain

$$\sum_{i=1}^m \alpha_i^u \left\{ \sqrt{(x_j - x_i)^2 + (y_j - y_i)^2 + c^2} \right\} = g(x_j, y_j) \tag{24}$$

$$j = n + 1, \dots, m$$

Solving  $m \times m$  system of Eqs. (23) and (24) by iteration, we obtain the coefficients  $\alpha_i^u$ . After that,  $u$  is found from Eq. (18) for all points located in the interior domain or on the boundary.

#### 3.2. Three-dimensional numerical formulation

In a similar way, we consider the 3D Poisson's equation

$$\nabla^2 \Phi_k = f_k(x, y, z) \quad (x, y, z) \in \text{domain} \tag{25}$$

$$\Phi_k = g_k(x, y, z) \quad (x, y, z) \in \text{boundary} \quad (26)$$

Let

$$\Phi_k = \sum_{i=1}^m \alpha_i \sqrt{r_i^2 + c^2} \quad (27)$$

where  $r_i$  is the Euclidean distance between the space point  $(x, y, z)$  and collocation point  $(x_i, y_i, z_i)$ , and  $c$  is a free shape parameter to be determined.  $f_k(x, y, z)$  is equal to  $-\nabla \times \vec{\omega}$  or  $0$ , and  $\Phi_k$  is equal to  $\vec{u}$  or  $\vec{\omega}$ . We choose  $m$  collocation points  $\{(x_i, y_i, z_i)\}_1^m$ . Now, we will illustrate Eq. (10) for  $u$ -velocity only. The other unknowns,  $v, w, \xi, \eta,$  and  $\zeta$  can be obtained in a similar way.

Let

$$u = \sum_{i=1}^m \alpha_i^u \sqrt{(x-x_i)^2 + (y-y_i)^2 + (z-z_i)^2 + c^2} \quad (28)$$

$$\zeta = \sum_{i=1}^m \alpha_i^\zeta \sqrt{(x-x_i)^2 + (y-y_i)^2 + (z-z_i)^2 + c^2} \quad (29)$$

$$\begin{aligned} \frac{\partial^2}{\partial x^2} \sum_{i=1}^m \alpha_i^u \sqrt{(x-x_i)^2 + (y-y_i)^2 + (z-z_i)^2 + c^2} \\ = \sum_{i=1}^m \alpha_i^u \frac{(y-y_i)^2 + (z-z_i)^2 + c^2}{(\sqrt{(x-x_i)^2 + (y-y_i)^2 + (z-z_i)^2 + c^2})^3} \end{aligned} \quad (30)$$

Similarly,

$$\begin{aligned} \frac{\partial^2}{\partial y^2} \sum_{i=1}^m \alpha_i^u \sqrt{(x-x_i)^2 + (y-y_i)^2 + (z-z_i)^2 + c^2} \\ = \sum_{i=1}^m \alpha_i^u \frac{(x-x_i)^2 + (z-z_i)^2 + c^2}{(\sqrt{(x-x_i)^2 + (y-y_i)^2 + (z-z_i)^2 + c^2})^3} \end{aligned} \quad (31)$$

$$\begin{aligned} \frac{\partial^2}{\partial z^2} \sum_{i=1}^m \alpha_i^u \sqrt{(x-x_i)^2 + (y-y_i)^2 + (z-z_i)^2 + c^2} \\ = \sum_{i=1}^m \alpha_i^u \frac{(x-x_i)^2 + (y-y_i)^2 + c^2}{(\sqrt{(x-x_i)^2 + (y-y_i)^2 + (z-z_i)^2 + c^2})^3} \end{aligned} \quad (32)$$

For vorticity derivative of  $\zeta$ ,

$$\begin{aligned} \frac{\partial}{\partial y} \sum_{i=1}^m \alpha_i^\zeta \sqrt{(x-x_i)^2 + (y-y_i)^2 + (z-z_i)^2 + c^2} \\ = \sum_{i=1}^m \alpha_i^\zeta \frac{y-y_i}{\sqrt{(x-x_i)^2 + (y-y_i)^2 + (z-z_i)^2 + c^2}} \end{aligned} \quad (33)$$

Similarly

$$\begin{aligned} \frac{\partial}{\partial z} \sum_{i=1}^m \alpha_i^\eta \sqrt{(x-x_i)^2 + (y-y_i)^2 + (z-z_i)^2 + c^2} \\ = \sum_{i=1}^m \alpha_i^\eta \frac{z-z_i}{\sqrt{(x-x_i)^2 + (y-y_i)^2 + (z-z_i)^2 + c^2}} \end{aligned} \quad (34)$$

Substituting Eqs. (30)–(34) into Eq. (10), and using the collocation method, we obtain  $n$  equations from the internal points and  $(m-n)$  equations from the boundary points. For the internal points,

$$\begin{aligned} \sum_{i=1}^m \alpha_i^u \left( \frac{2(x_j-x_i)^2 + 2(y_j-y_i)^2 + 2(z_j-z_i)^2 + 3c^2}{(\sqrt{(x_j-x_i)^2 + (y_j-y_i)^2 + (z_j-z_i)^2 + c^2})^3} \right) \\ = - \left[ \sum_{i=1}^m \alpha_i^\zeta \frac{(y_j-y_i)}{(\sqrt{(x_j-x_i)^2 + (y_j-y_i)^2 + (z_j-z_i)^2 + c^2})} \right. \\ \left. - \sum_{i=1}^m \alpha_i^\eta \frac{(z_j-z_i)}{(\sqrt{(x_j-x_i)^2 + (y_j-y_i)^2 + (z_j-z_i)^2 + c^2})} \right] \\ j = 1, \dots, n \end{aligned} \quad (35)$$

Similarly for the  $v$  and  $w$  components, we find

$$\begin{aligned} \sum_{i=1}^m \alpha_i^v \left( \frac{2(x_j-x_i)^2 + 2(y_j-y_i)^2 + 2(z_j-z_i)^2 + 3c^2}{(\sqrt{(x_j-x_i)^2 + (y_j-y_i)^2 + (z_j-z_i)^2 + c^2})^3} \right) \\ = - \left[ \sum_{i=1}^m \alpha_i^\xi \frac{(z_j-z_i)}{(\sqrt{(x_j-x_i)^2 + (y_j-y_i)^2 + (z_j-z_i)^2 + c^2})} \right. \\ \left. - \sum_{i=1}^m \alpha_i^\zeta \frac{(x_j-x_i)}{(\sqrt{(x_j-x_i)^2 + (y_j-y_i)^2 + (z_j-z_i)^2 + c^2})} \right] \\ j = 1, \dots, n \end{aligned} \quad (36)$$

$$\begin{aligned} \sum_{i=1}^m \alpha_i^w \left( \frac{2(x_j-x_i)^2 + 2(y_j-y_i)^2 + 2(z_j-z_i)^2 + 3c^2}{(\sqrt{(x_j-x_i)^2 + (y_j-y_i)^2 + (z_j-z_i)^2 + c^2})^3} \right) \\ = - \left[ \sum_{i=1}^m \alpha_i^\eta \frac{(x_j-x_i)}{(\sqrt{(x_j-x_i)^2 + (y_j-y_i)^2 + (z_j-z_i)^2 + c^2})} \right. \\ \left. - \sum_{i=1}^m \alpha_i^\xi \frac{(y_j-y_i)}{(\sqrt{(x_j-x_i)^2 + (y_j-y_i)^2 + (z_j-z_i)^2 + c^2})} \right] \\ j = 1, \dots, n \end{aligned} \quad (37)$$

We choose  $(m-n)$  boundary points  $\{(x_i, y_i, z_i)\}_{n+1}^m$ . From Eq. (26), we obtain

$$\begin{aligned} \sum_{i=1}^m \alpha_i^k \left\{ \sqrt{(x_j-x_i)^2 + (y_j-y_i)^2 + (z_j-z_i)^2 + c^2} \right\} \\ = g_k(x_j, y_j, z_j) \end{aligned} \quad (38)$$

$$j = n+1, \dots, m, \quad k = u, v, w$$

Solving the  $m \times m$  system of Eqs. (35)–(38) by iteration, we obtain  $\alpha_i^k$ , the velocity coefficients. Similarly, we use the same method to solve Eqs. (7)–(9) for the vorticity transfer

equation, and then get the coefficient  $\alpha_i^n$  of the vorticity, where  $n$  is chosen from  $\xi, \eta$ , and  $\zeta$ .

It is noticed that in order to reduce the size of the coupling of velocity and vorticity fields, we choose the iterative scheme to solve the velocity and vorticity in sequence. The vorticity provides the internal source for the velocity; while the solution of vorticity needs the boundary condition for vorticity from which it is obtained by taking the curl of the velocity field. The details are elaborated in Section 4.

#### 4. Solution procedures

The MQ procedure is applied to solve the Stokes flows via the velocity–vorticity formulation. In most incompressible viscous flow problems, the essential boundary conditions are the prescribed velocities. The vorticity boundary conditions are not known a priori in general, and should be determined iteratively from the computation of the velocity field. In the present model, the velocity Poisson equations are firstly solved to obtain the velocity and the associated vorticity boundary conditions. After that, the vorticity is obtained by solving the vorticity Laplace equation. The solution procedures are described in the following steps

1. Solve the velocity Poisson equations using the MQ (Kansa's) method by taking known vorticity components.
2. Calculate the unknown boundary velocity or velocity flux values.
3. Calculate the velocity distribution and velocity derivatives at all nodal points.
4. Determine the new vorticity boundary values using the definition of vorticity.
5. Solve the vorticity Laplace equation using the MQ (Kansa's) method.
6. Calculate the unknown vorticity values throughout the domain.
7. Calculate the derivatives of the vorticity vectors to be used in the velocity Poisson equations.
8. Check for the convergence of the velocity and vorticity components in the present iteration, for example,

$$\left| u_i^{k+1} - u_i^k \right| \leq 10^{-4}$$

9. If convergence criterion is satisfied, then stop. Otherwise replace the old vorticity by a new one and then go to step 1.

The equations of the full matrix are then solved by using the Gaussian elimination technique. We have encountered no ill-conditioning problem of the full matrix as big as the size of 3375 nodes without the introduction of precondition

or domain decomposition methods as commonly adopted in the literature.

#### 5. Numerical examples

The present MQ scheme has been applied to test three problems to verify the feasibility and accuracy of the method. The test problems are the classical 'driven flow in a square cavity' and 'flow in a circular cavity' problems for which many analytical and numerical model results are available in the literature for 2D Stokes flows. A cubic cavity flow on the other hand is employed to illustrate the feasibility of the 3D Stokes flows.

**Example 1.** The first model problem consists of a 2D square cavity filled with incompressible viscous fluid moving at a constant velocity on the top lid. No slip or impervious boundary conditions are imposed on any wall, with the velocity at the upper wall set to unity. For the present computation, 33 boundary points and 88 internal points are used. In this example, we use  $11 \times 11$  nodes to collocate the region,  $(1,0) \times (0,1)$ . The computational domain is divided into two regions, boundary and interior regions. The value of the shape function  $c$  is taken as 0.39.

The square cavity flow with configuration and boundary conditions is illustrated in Fig. 1. The velocity vector is shown in Fig. 2, and the distribution of vorticity is depicted in Fig. 3, respectively. We use two sets of collocation points for  $11 \times 11$  nodes and  $13 \times 13$  nodes respectively to express the  $x$ -component velocity profile ( $u$ ) on the vertical centerline of the cavity, and compared with the accuracy of the two collocation sets. The diagram of Fig. 4 reveals almost no difference between the two collocations for the velocity

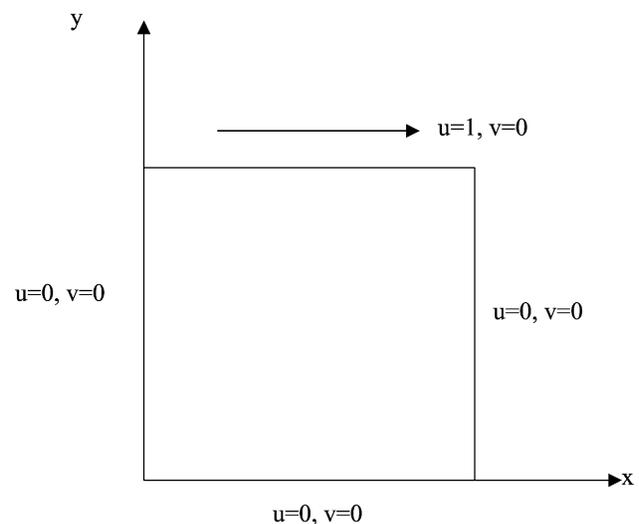


Fig. 1. Square cavity flow problem with boundary conditions.

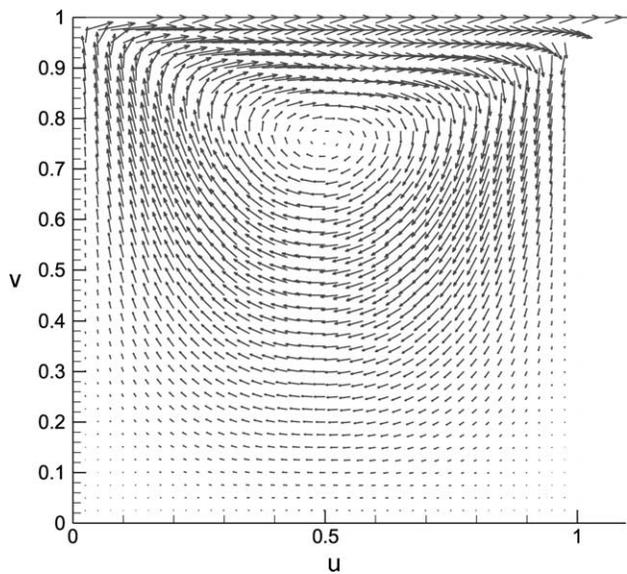


Fig. 2. Velocity vectors for Stokes flow in a square cavity.

profiles. Fig. 5 displays the velocity profile ( $v$ ) of the  $y$ -component on the horizontal centerline for steady-state Stokes flow by  $11 \times 11$  collocation points, and Fig. 6 exhibits the velocity profile ( $u$ ) of the  $x$ -component on the vertical centerline of the cavity. The results are compared with Burggraf series solution [16], a FEM solution [17], a BEM solution [18], and a DRBEM (direct method) solution [19]. Good agreement is observed even in such small sets of collocation points. As can be seen in Fig. 6, the present MQ (Kansa's) method is as accurate as the other numerical methods, even if very few collocation points are used.

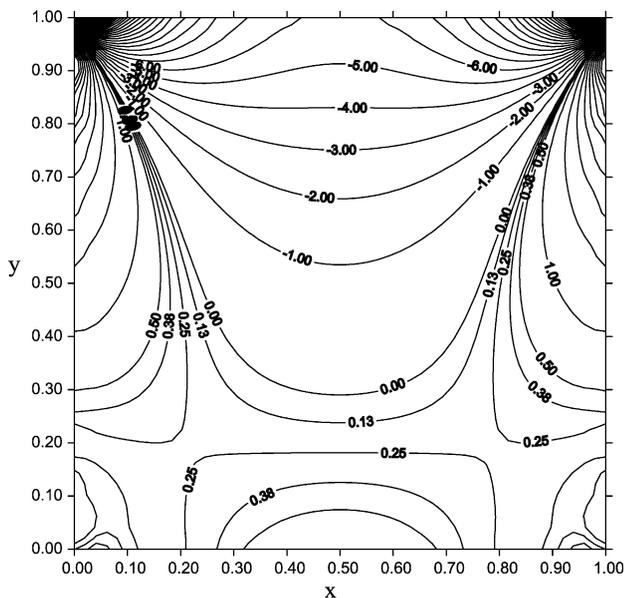


Fig. 3. Vorticity distribution for Stokes flow in a square cavity.

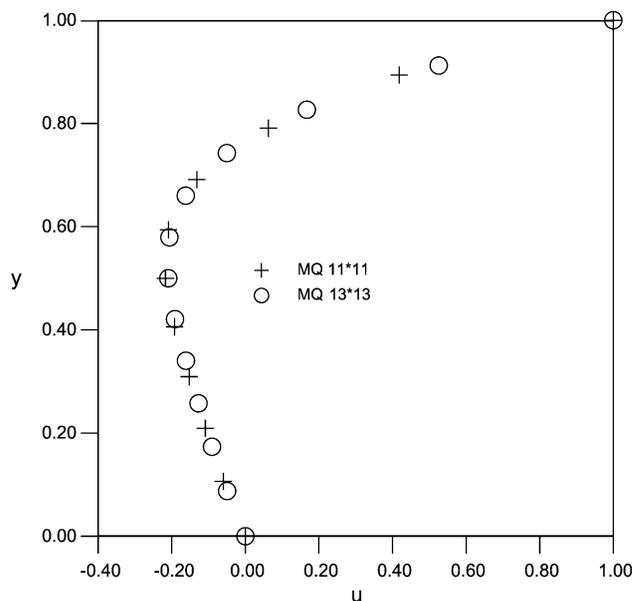


Fig. 4. Velocity profile ( $u$ ) on the centerline at  $x = 0.5$  of a square cavity; comparison of mesh convergence.

**Example 2.** The second model problem consists of a recirculating flow in a 2D circular cavity [16,20]. The radius of the circular cavity is assumed to be unity. The configuration and boundary conditions of the problem is depicted in Fig. 7. In the upper half of the boundary, the velocity  $u_\theta = 1$  in a contour clockwise sense and  $u_r = 0$  are prescribed, and in the lower half,  $u_\theta = u_r = 0$  are imposed. For this computation, a total of 612 boundary points and internal points are used in this study.

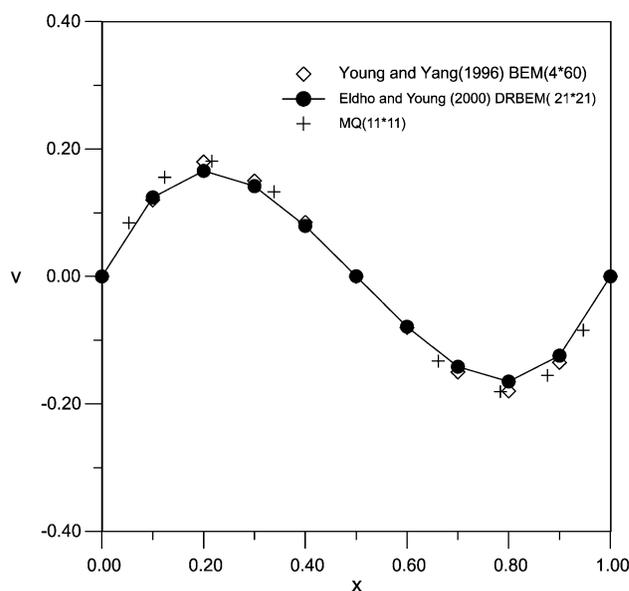


Fig. 5. Velocity profile ( $v$ ) on the centerline at  $y = 0.5$  of a square cavity.

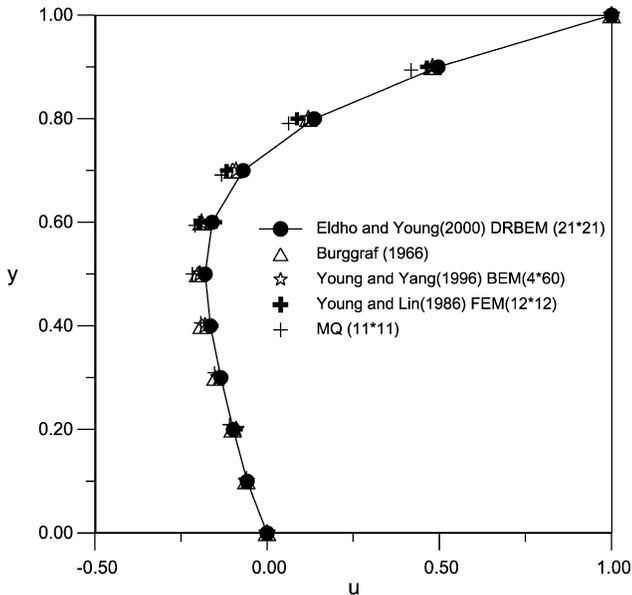


Fig. 6. Velocity profile ( $u$ ) on the centerline at  $x = 0.5$  of a square cavity.

The velocity vector is observed in Fig. 8, and the distribution of vorticity contour is shown in Fig. 9. The present computations give very good results as compared with the other solutions. Fig. 10 depicted the  $x$ -component velocity profile ( $u$ ) on the vertical centerline, and Fig. 11 describes the  $y$ -component velocity profile ( $v$ ) on the horizontal centerline. The results are compared with Burggraf series solution [16] and Hwu et al. analytical solution [20], and both also show good agreements.

**Example 3.** The third model problem is a 3D cubic cavity, with the top plate moving with unit velocity in the  $x$ -direction ( $z = 1, 0 \leq x \leq 1, 0 \leq y \leq 1, u = 1$ ). Others are set with no-slip and impervious Dirichlet conditions.

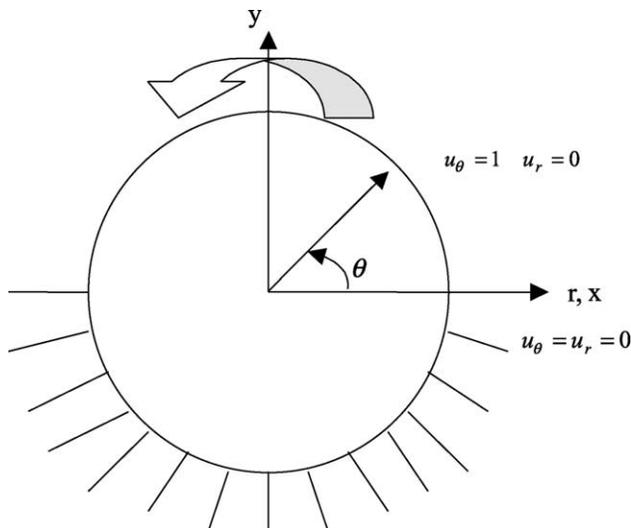


Fig. 7. Circular cavity flow problem with boundary conditions.

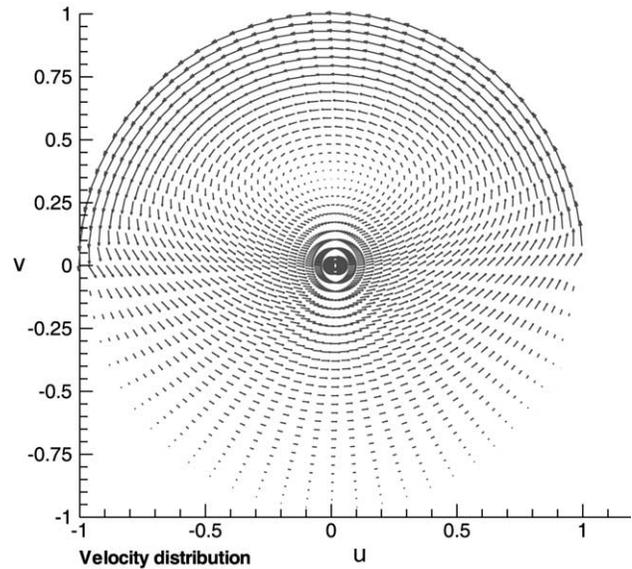


Fig. 8. Velocity vectors for Stokes flow in a circular cavity.

The configuration and boundary conditions of the problem are shown in Fig. 12. We used regular collocation points in the cube for three different uniform meshes of size  $11 \times 11 \times 11$ ,  $13 \times 13 \times 13$ , and  $15 \times 15 \times 15$  in this simulation. A mesh sensitivity study using the three meshes of size  $11 \times 11 \times 11$ ,  $13 \times 13 \times 13$ , and  $15 \times 15 \times 15$  indicates that the mesh  $15 \times 15 \times 15$  predicts velocity and vorticity with the acceptable error.

The velocity vectors which take  $x$ -component velocity  $u$  and  $z$ -component velocity  $w$  are shown in Fig. 13. The distribution of vorticity which takes the  $x-z$  plane at  $y = 0.5$  is depicted in Fig. 14. From another aspect, velocity vector

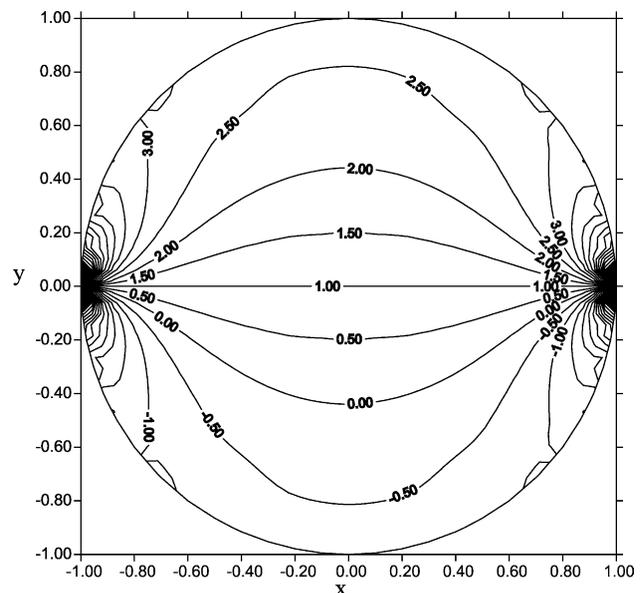


Fig. 9. Vorticity distribution for Stokes flow in a circular cavity.

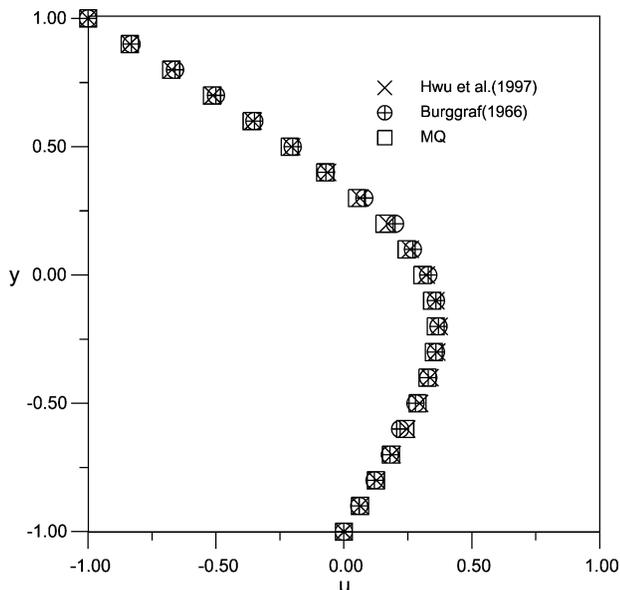


Fig. 10. Comparison of  $u$ -velocity profile at  $x = 0$  for a circular cavity.

at  $x$ - $y$  plane at  $z = 0.5$  for Stokes flow in the cubic cavity is displayed in Fig. 15. The velocity affects the distribution of the intensity of the vorticity, so we look more closely at the vorticity profile for more cross sections. Fig. 16 expresses the  $\xi$  vorticity intensity which is located at  $z = 0.5$  of  $x$ - $y$  plane; the results indicate the symmetric characteristics with respect to  $x$ - and  $y$ -directions. The effect of the wall makes vorticity distribution more concentrated on the edges of the cavity. This consequence is conspicuously revealed in Figs. 17 and 18, respectively. Fig. 19 shows the  $x$ -component velocity profile ( $u$ ) on the vertical centerline of the cavity, and Fig. 20 shows the  $z$ -component velocity profile ( $w$ ) on the horizontal

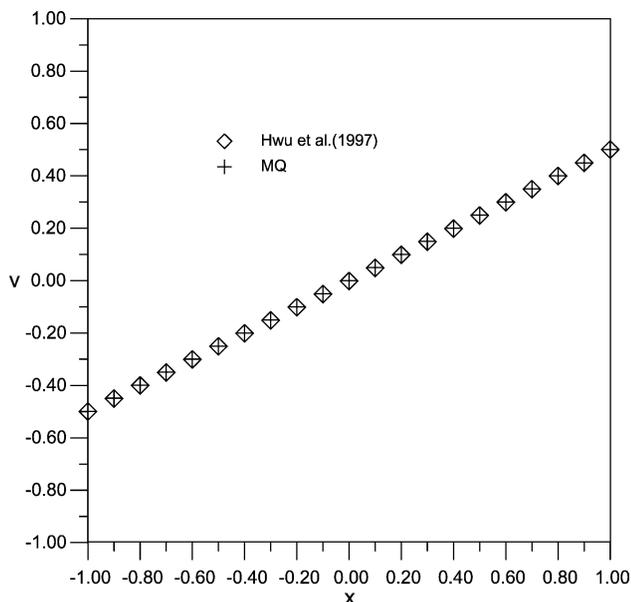


Fig. 11. Comparison of  $v$ -velocity profile at  $y = 0$  for a circular cavity.

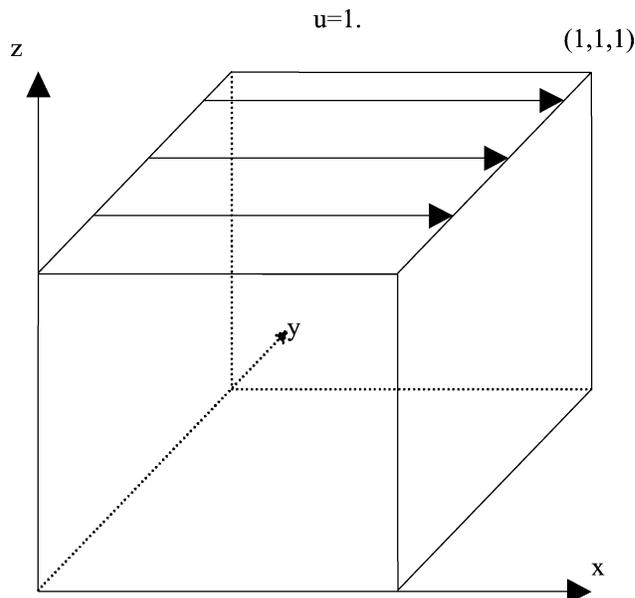


Fig. 12. Cubic cavity flow problem with boundary conditions.

centerline for steady-state Stokes flow of the cubic cavity. The results are compared with those obtained using meshless BEM method (Tsai et al. [21]) and the results of a traditional BEM-FEM (Young et al. [22]). The result of our present method with  $11 \times 11 \times 11$  mesh is not as good as the results obtained by the meshless BEM (Tsai et al. [21]). However, it is conjectured that adding more point density, for example by using meshes of size  $13 \times 13 \times 13$  or  $15 \times 15 \times 15$ , the quality of our results could be improved, and also can capture the effect of wall at driven direction with increased node density. Nevertheless, the MQ (Kansa's) scheme is very easy to implement as compared to other numerical methods.

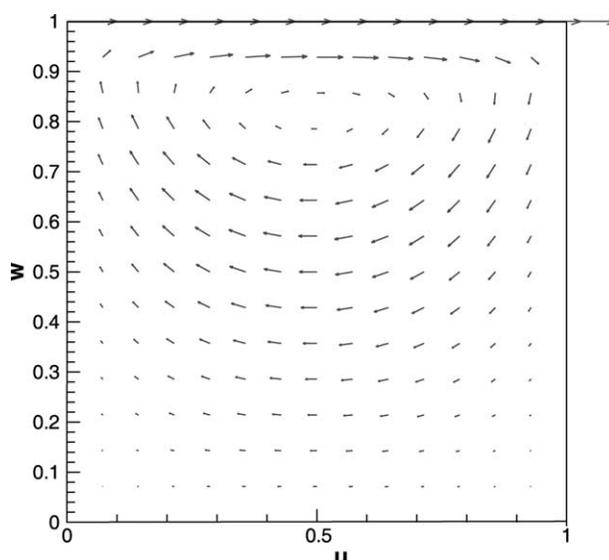


Fig. 13. Velocity vectors of  $x$ - $z$  plane at  $y = 0.5$  for Stokes flow in a cubic cavity.

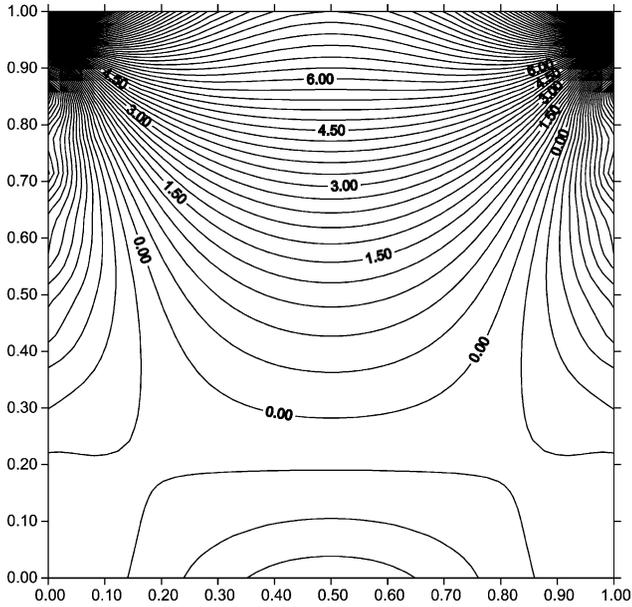


Fig. 14.  $\eta$ -component vorticity distribution of  $x$ - $z$  plane at  $y = 0.5$  for Stokes flow in a cubic cavity.

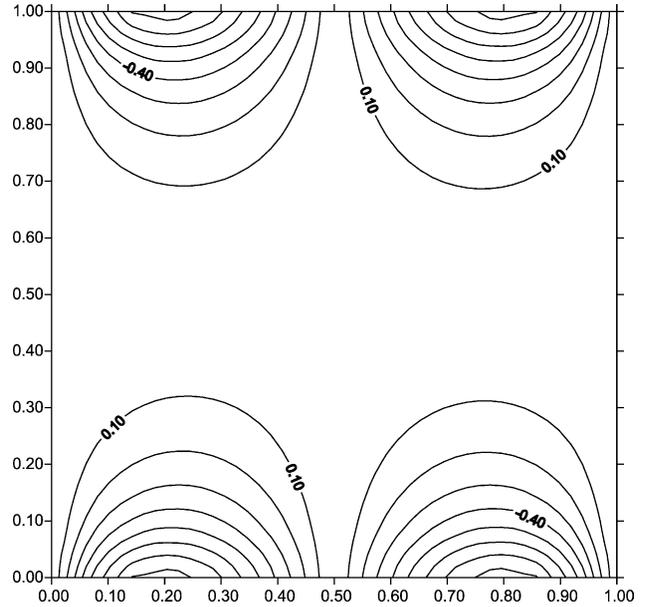


Fig. 16.  $\xi$ -component vorticity distribution of  $x$ - $y$  plane at  $z = 0.5$  for Stokes flow in cubic cavity.

In order to make more quantitative comparisons with analytic or other numerical solutions, the following criterion of root mean squared (RMS) error analysis is established. The RMS errors of  $x$ -component velocity profile ( $u$ ) on the vertical centerline, as well as  $y$ -component velocity profile ( $v$ ) on the horizontal centerline are both considered. The base used for the square cavity is the Burggraf series solution [16], while the analytic solution of Hwu et al. [20] is employed for the circular cavity. The RMS errors for 2D square cavity and circular

cavity are shown in Table 1. The magnitudes are in the range of  $8 \times 10^{-3}$  to  $4 \times 10^{-4}$ . For the 3D cubic cavity flow problem, the base used for calculating RMS is Tsai et al. [21] meshless results since there are no analytic solutions available. The corresponding RMS error is illustrated in Table 2. The values lie between  $6 \times 10^{-2}$  and  $8 \times 10^{-8}$ . In general, the present MQ (Kansa's) method renders reasonable and comparable accuracy as compared with exact and other numerical schemes.

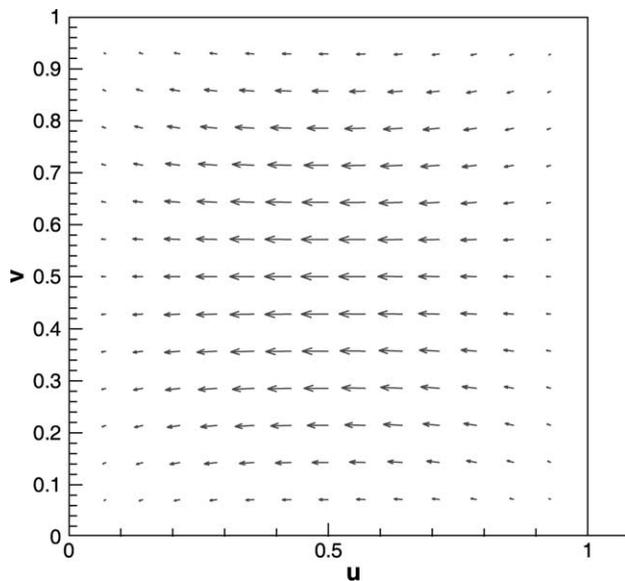


Fig. 15. Velocity vectors of  $x$ - $y$  plane at  $z = 0.5$  for Stokes flow in a cubic cavity.

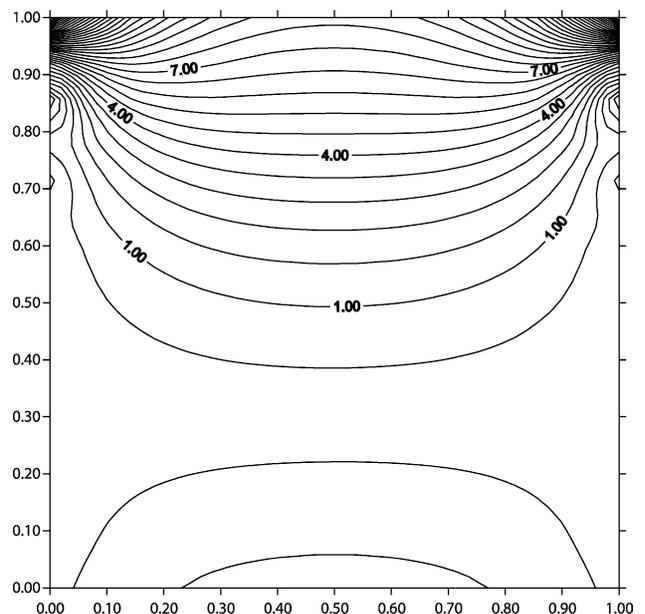


Fig. 17.  $\eta$ -component vorticity distribution of  $y$ - $z$  plane at  $x = 0.5$  for Stokes flow in cubic cavity.

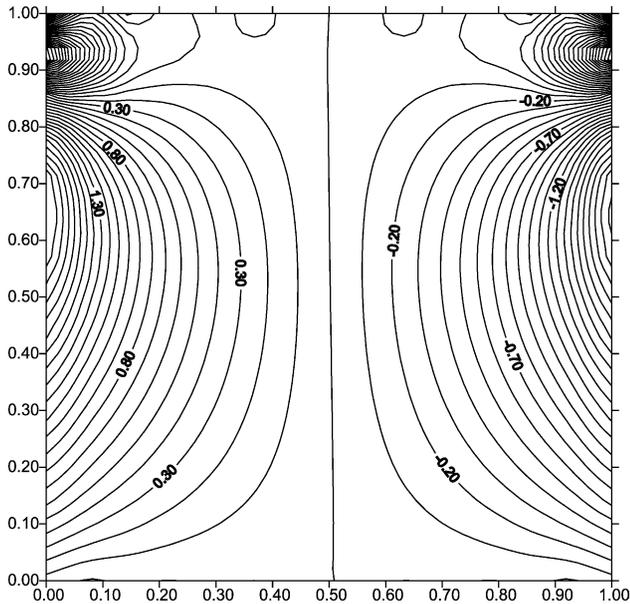


Fig. 18.  $\zeta$ -component vorticity distribution of  $y$ - $z$  plane at  $x = 0.5$  for Stokes flow in cubic cavity.

6. Conclusions

Stokes flows are investigated in 2D square and 2D circular cavity and a 3D cubic cavity. In comparing with analytical solutions and other numerical solutions, the results show that the present scheme is efficient and accurate for using very few collocation points with the merits of meshfree and adaptive mesh advantages. In this investigation, the velocity–vorticity formulation is employed. It mainly deals with the coupled calculation of the vorticity and velocity equations. It means that the accuracy of both results depends on the mesh size of each

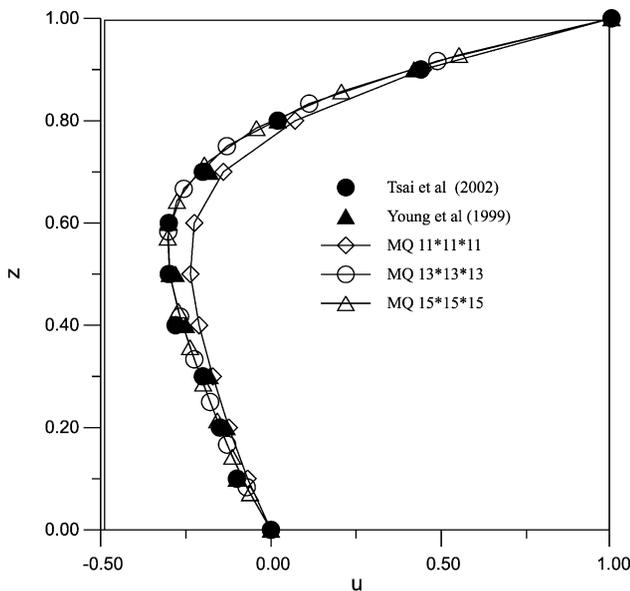


Fig. 19. Velocity profile ( $u$ ) on the centerline at  $x = 0.5$  of a cubic cavity.

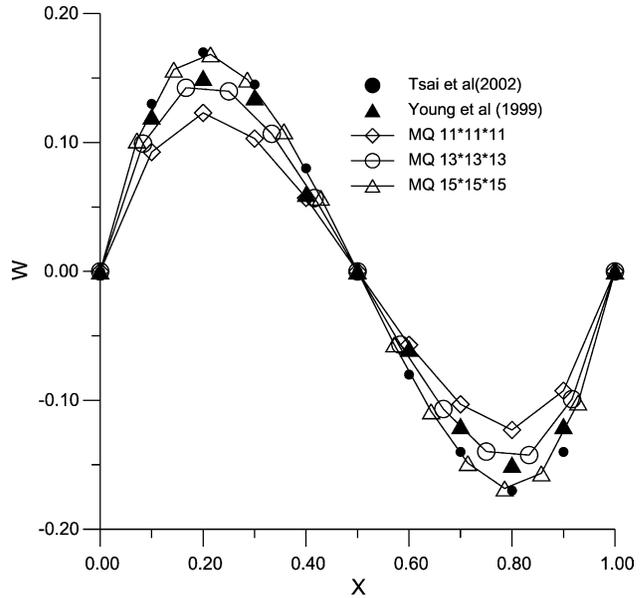


Fig. 20. Velocity profile ( $w$ ) on the centerline at  $y = 0.5$  of a cubic cavity.

Table 1  
The RMS error for square and circular cavities

Shape	Variable		
	Nodes	$u$	$v$
Square	121	0.00299	0.000492
Circular	612	0.00882	0.00116

other, so we can increase the number of points at some high gradient places, such as near the singular points to raise the resolution of the vorticity accuracy.

The demonstrated good performance of the MQ (Kansa’s) scheme shows that it is a powerful tool for the numerical solution of incompressible viscous fluid flows. Although 2D and 3D Stokes flow problems are primarily solved, we expect the MQ (Kansa’s) scheme will have great potential to solve other 2D and 3D incompressible viscous flow problems that are currently under study.

The main limitation of the MQ scheme is the huge storage of data coming from the operator matrix. If the numerical model is extended to 3D Navier–Stokes problems, the storage of data would become very much larger than that of the 2D model. Therefore, it deserves further research to overcome this problem. Finally, the MQ

Table 2  
The RMS error for cubic cavity

Mesh size	Variable	
	$u$	$w$
$11 \times 11 \times 11$	0.0583	0.00698
$13 \times 13 \times 13$	0.0177	0.00643
$15 \times 15 \times 15$	0.00943	$7.94 \times 10^{-08}$

(Kansa's) is worthy to investigate the more challenging 2D and 3D Navier–Stokes equations especially for the higher Reynolds number flows.

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