

Dynamic Crack Propagation in a Layered Medium Under Antiplane Shear

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In this study, the transient analysis of dynamic antiplane crack propagation with a constant velocity in a layered medium is investigated. The individual layers are isotropic and homogeneous. Infinite numbers of reflected cylindrical waves, which are generated from the interface of the layered medium, will interact with the propagating crack and make the problem extremely difficult to analyze. A useful fundamental solution is proposed in this study, and the solution can be determined by superposition of the fundamental solution in the Laplace transform domain. The proposed fundamental problem is the problem of applying exponentially distributed traction (in the Laplace transform domain) on the propagating crack faces. The Cagniard's method for Laplace inversion is used to obtain the transient solution in time domain. The exact closed-form transient solutions of dynamic stress intensity factors are expressed in compact formulations. These solutions are valid for an infinite length of time and have accounted for contributions from all the incident and reflected waves interaction with the moving crack tip. Numerical results of dynamic stress intensity factors for the propagation crack in layered medium are evaluated and discussed in detail.

1 Introduction

Most of the analyses done regarding cracked bodies are quasi-static. There are numerous situations that the material inertia becomes significant and must be taken into account in the analyses. The question of whether or not inertial effects are significant depends on the loading conditions and the geometrical configuration of the body. Inertial effects can arise either from applying dynamic loading on a cracked solid or from rapid crack propagation. The inherent time dependence of a dynamic fracture process results in mathematical models that are more complex than equivalent quasi-static models. However, there is substantial interest in the dynamic fracture problem due to its importance in many engineering applications. The relevant applications are NDE of layered media, rupture due to earthquakes in earth's crust, plate tectonics, composite materials, and functionally graded materials.

The main purpose in solving problems concerned with dynamic crack propagation is to determine the dependence of the crack-tip field characterizing parameters on the applied loading and on the configuration of the body. The investigation of a propagation crack in a brittle solid began with the pioneering analysis of Yoffe (1951). She considered a steady-state crack growth problem of a crack of fixed length propagating in an infinite elastic body subjected to a uniform remote tensile loading normal to the crack line. Although this is a physically unrealistic problem, it did at least provide an indication of the influence of crack speed on the stress state of a rapidly propagating crack. Craggs (1960) considered a semi-infinite crack extending at constant speed, with the crack face loading moving with the same speed as the crack tip in such a way that the entire deformation field is constant as seen by an observer moving with the

crack tip. The self-similar dynamic crack propagation problem was contributed by Broberg (1960) who was among the first to present detailed analyses of crack propagation as a transient process. He solved the dynamic problem of a crack that suddenly grows from zero length at a constant speed. Baker (1962) subsequently generalized Broberg's solution to include a finite initial crack. Although the above mentioned artificial solutions have no direct application, they also have provided useful insights and continue to be used today for the assessment of dynamic numerical analyses.

In a series of papers, Freund (1972a, 1972b, 1973, 1974) developed important analytical methods for evaluation of the transient stress field of a propagating crack in a two-dimensional geometric configuration under quite general dynamic loading situation. These particular cases analyzed by Freund are also self-similar, but they are solved by means of integral transform methods rather than by direct application to similarity arguments. An indirect analytical approach proposed by Freund based on superposition over a fundamental solution, which opens a way for analysis of certain problems of crack propagation at nonuniform speed. Based on the superposition method proposed by Freund, a series of problems for nonplanar crack propagation in an infinite domain was solved by Ma and Burgers (1986, 1987, 1988) and Ma (1988, 1990). The structure of the near-tip field of crack propagation at nonuniform speeds was discussed in detail by Freund and Rosakis (1992). A representation of the crack-tip field was obtained in the form of an expansion about the crack tip in powers of radial coordinate, with the coefficients depending on the time rates of change of crack-tip speed and stress intensity factor. This representation was used to interpret some experimental observations and some estimates were made of the practical limits of using a stress intensity factor field alone to characterize the local fields. A thorough summary of the application of the main direct methods of analysis for transient problem in dynamic fracture has been given by Freund (1990).

Most of the solved dynamic fracture problems are regarded as a crack subjected to applying a uniformly distributed dynamic loading on the crack faces or subjected to incident plane waves. For the above-mentioned problems, either the direct application

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transform on time t . The one-sided Laplace transform with respect to time and the two-sided Laplace transform with respect to ξ are defined by

$$\bar{w}(\xi, y, s) = \int_0^{\infty} w(\xi, y, t) e^{-st} dt,$$

$$\tilde{w}(\lambda, y, s) = \int_{-\infty}^{\infty} \bar{w}(\xi, y, s) e^{-s\lambda\xi} d\xi.$$

This fundamental problem can be solved by using the standard transform method and the Wiener-Hopf technique. The governing equation can be represented by the two-dimensional wave equation

$$\frac{\partial^2 w}{\partial x^2} + \frac{\partial^2 w}{\partial y^2} = b^2 \frac{\partial^2 w}{\partial t^2}, \quad (2.3)$$

where b is the slowness of the shear wave and is given by

$$b = 1/v_s = \sqrt{\rho/\mu},$$

in which $w(x, y, t)$ is the displacement normal to the xy -plane; v_s is the shear wave speed, μ and ρ are the respective shear modulus and the mass density of the material. The nonvanishing shear stresses are

$$\tau_{yz} = \mu \frac{\partial w}{\partial y}, \quad \tau_{xz} = \mu \frac{\partial w}{\partial x}. \quad (2.4)$$

In analyzing this problem, it is convenient to express the governing equation in the moving coordinates $\xi - y$ as follows:

$$(1 - b^2 v^2) \frac{\partial^2 w}{\partial \xi^2} + \frac{\partial^2 w}{\partial y^2} + 2(b^2 v) \frac{\partial^2 w}{\partial \xi \partial t} - b^2 \frac{\partial^2 w}{\partial t^2} = 0.$$

This fundamental problem can be solved by the application of integral transforms. Applying the one-sided Laplace transform over time, the two-sided Laplace transform over ξ under the restriction of $\text{Re}(\eta) > \text{Re}(\lambda)$, finally the Wiener-Hopf technique is implemented. The solutions of stresses and displacement in the transform domain for the boundary conditions (2.1) and (2.2) are

$$\bar{\tau}_{yz}(\xi, y, s) = \frac{1}{2\pi i} \int_{\Gamma_\lambda} \frac{\alpha_\mp^*(\lambda) e^{-s(\alpha^* y - \lambda \xi)}}{\alpha_\mp^*(\eta)(\eta - \lambda)} d\lambda, \quad (2.5)$$

$$\bar{\tau}_{xz}(\xi, y, s) = -\frac{1}{2\pi i} \int_{\Gamma_\lambda} \frac{\lambda e^{-s(\alpha^* y - \lambda \xi)}}{\alpha_\mp^*(\eta)(\eta - \lambda) \alpha_\mp^*(\lambda)} d\lambda, \quad (2.6)$$

$$\bar{w}(\xi, y, s) = -\frac{1}{2\pi i} \int_{\Gamma_\lambda} \frac{e^{-s(\alpha^* y - \lambda \xi)}}{\mu s \alpha_\mp^*(\eta)(\eta - \lambda) \alpha_\mp^*(\lambda)} d\lambda, \quad (2.7)$$

where

$$\alpha^*(\lambda) = \sqrt{b + \lambda(1 - bv)} \sqrt{b - \lambda(1 + bv)} = \alpha_\mp^*(\lambda) \alpha_\pm^*(\lambda).$$

The corresponding result of the dynamic stress intensity factor in the Laplace transform domain is

$$\begin{aligned} \bar{K}(s) &= \lim_{\xi \rightarrow 0} \sqrt{2\pi\xi} \bar{\tau}_{yz}(\xi, 0, s) \\ &= -\frac{\sqrt{2}\sqrt{1 - bv}}{\sqrt{s} \alpha_\mp^*(\eta)}. \end{aligned} \quad (2.8)$$

3 Transient Analysis for Propagating Crack in a Layered Media

Consider the problem of a semi-infinite crack in an infinitely long strip with finite width $2l$ of a layered medium as shown in Fig. 1. At time $t = 0$, a pair of equal and opposite concentrated antiplane dynamic loadings with magnitude p are applied at the crack faces with a distance h from the crack tip, the crack starts

to propagate with constant velocity v at a delay time that the wave reach the crack tip (i.e., $t_f = b/h$) in the x -direction along the symmetry line $y = 0$ of the strip. The time dependence of the loading is represented by the Heaviside step function $H(t)$. In this problem, the cylindrical wave induced from the dynamic concentrated loading and the diffracted waves generated from the propagating crack tip will be reflected from horizontal interface boundaries, these waves will interact with the propagating crack at some later time. The major difficulty in analyzing this problem will be the one that deals with the interaction of reflected waves with the propagating crack and the superposition technique of the fundamental solutions in the Laplace transform domain will be used in the analysis. The transient solutions are composed of incident field, reflected field and diffracted field, which will be denoted by superscripts of i , r , and d , respectively. Before the time that the i and d waves reflected from the interface boundaries of the strip, the problem can be considered as a semi-infinite crack propagating in an unbounded medium. In the following analysis, the solution is valid for the case that crack starts to propagating before the waves (i.e., i wave and d wave) generated from the stationary crack returns to the stationary crack (i.e., $t_f < 2bl$) and $b_1 < b_2$.

The incident field of the cylindrical wave generated by the concentrated loading expressed in the Laplace transform domain can be obtained as follows:

$$\bar{\tau}_{yz}(x, 0, s) = \frac{1}{2\pi i} \int_{\Gamma_\lambda} -p e^{s\lambda(x+h)} d\lambda. \quad (3.1)$$

The applied traction on the crack face as indicated in (3.1), has a functional form $e^{s\lambda x}$. Since the solutions of applying traction $e^{s\eta x}$ on crack faces in the Laplace transform domain have been solved in Section 2 by setting $v = 0$, the diffracted field generated from the stationary semi-infinite crack can be constructed by superimposing the incident wave traction that is equal to (3.1). When we combine (2.5) (by setting $v = 0$) and (3.1), the solution of diffracted wave in the Laplace transform domain can be expressed as follows:

$$\begin{aligned} \bar{\tau}_{yz}^d(x, y, s) &= \frac{1}{2\pi i} \int_{\Gamma_\lambda} -p \left\{ \frac{1}{2\pi i} \int_{\Gamma_{\eta_2}} \frac{(b_1 + \eta_2)^{1/2}}{(\lambda - \eta_2)(b_1 + \lambda)^{1/2}} \right. \\ &\quad \left. \times e^{s\lambda h} e^{-s(\alpha_\pm y - \eta_2 x)} d\eta_2 \right\} d\lambda. \end{aligned} \quad (3.2)$$

By using the Cagniard-de Hoop method of Laplace inversion, the incident and diffracted stress fields in time domain for the stationary crack are obtained as follows:

$$\bar{\tau}_{yz}^i(x, y, t) = \frac{-pt \sin \theta}{\pi r (t^2 - b_1^2 r^2)^{1/2}} H(t - b_1 r), \quad (3.3)$$

$$\begin{aligned} \bar{\tau}_{yz}^d(x, y, t) &= \frac{p}{2\pi^2} \int_{b_1 h}^{t-b_1 r_2} \text{Re} \left[H(\eta_1^+, \eta_2^+) \frac{\partial \eta_1^+}{\partial t_1} \frac{\partial \eta_2^+}{\partial t_2} \right. \\ &\quad \left. - H(\eta_1^-, \eta_2^+) \frac{\partial \eta_1^-}{\partial t_1} \frac{\partial \eta_2^+}{\partial t_2} \right] dt_1, \end{aligned} \quad (3.4)$$

where

$$\eta_1^\pm = -\frac{t_1}{h} \pm ie,$$

$$\eta_2^\pm = -\frac{t_2}{r_2} \cos \theta_2 \pm i \left(\frac{t_2^2}{r_2^2} - b_1^2 \right)^{1/2} \sin \theta_2,$$

$$r = [(x+h)^2 + y^2]^{1/2}, \quad \theta = \cos^{-1} [(x+h)/r],$$

$$r_2 = (x^2 + y^2)^{1/2}, \quad \theta_2 = \cos^{-1} (x/r_2),$$

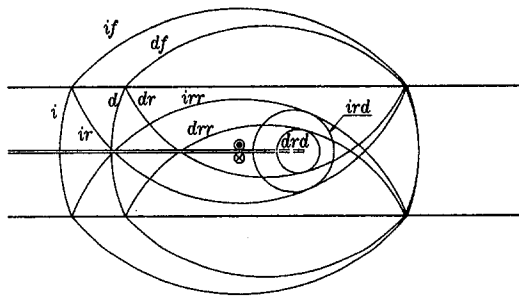


Fig. 2 Configuration of wave fronts in the layered medium after the crack propagates with constant velocity

$$H(\eta_1, \eta_2) = \frac{(b_1 + \eta_2)^{1/2}}{(\eta_1 - \eta_2)(b_1 + \eta_1)^{1/2}},$$

$$t = t_1 + t_2.$$

The corresponding dynamic stress intensity factor induced by diffracted wave will be

$$K^s(t) = p \sqrt{\frac{2}{\pi h}} H(t - b_1 h). \quad (3.5)$$

The dynamic stress intensity factor shown in (3.5) jumps from zero at the instant that the cylindrical wave generated by the concentrated loading reaches the crack tip. At time $t = t_f = b_1 h$, the dynamic stress intensity factor reaches its critical value and the crack starts to propagate with constant velocity v along the line $y = 0$. The transient full-field analysis for a propagating crack just mentioned above has also been solved by Ma and Chen (1992) by using the method proposed by Freund (1972b). In their investigations, the transient solution for constant-speed crack propagation is obtained by determining a fundamental solution for a concentrated force appearing through the moving crack tip, and then building up the general solution by superposition. In this study, a more direct and simple methodology will be used to solve this problem. We consider the transient problem of a semi-infinite crack propagates at $t = t_f = b_1 h$ with symmetric concentrated loading applied only on the original crack faces $-\infty < x < 0$. The applied symmetric concentrated loading p

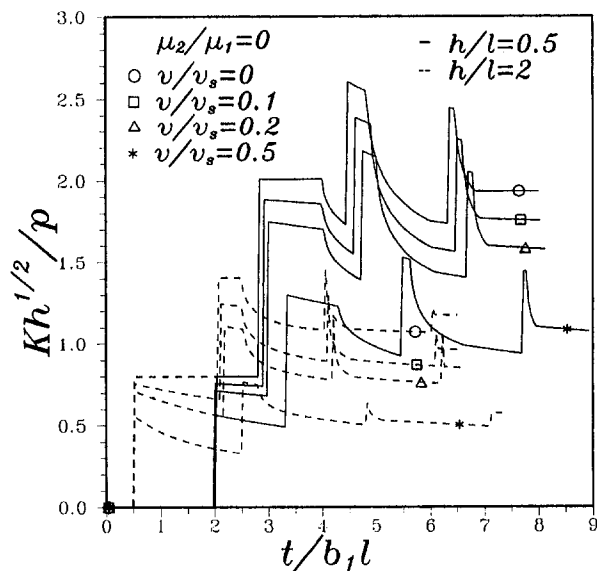


Fig. 3 Transient response of the dynamic stress intensity factor for various crack propagating speed

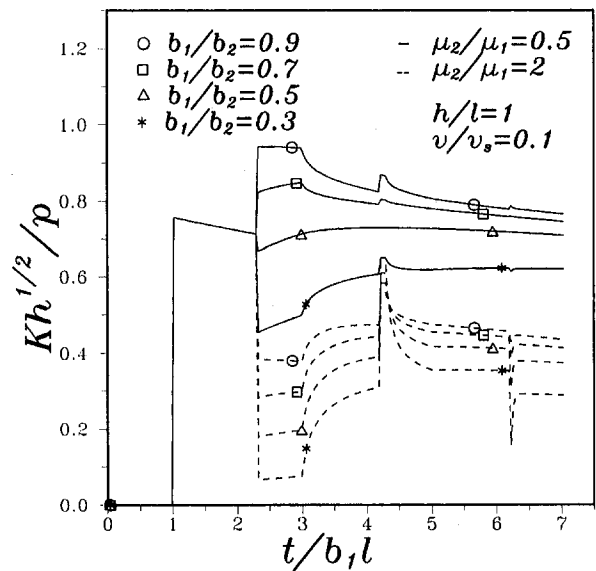


Fig. 4 Transient response of the dynamic stress intensity factor for different ratio of shear wave speed

on the original crack faces written in the Laplace transform domain for the moving coordinate system will have the following form:

$$\bar{\tau}_{yz}(\xi, 0, s) = \frac{1}{2\pi i} \int_{\Gamma_\lambda} \frac{pd}{(\lambda - d)} e^{sh(1-b_1v)\lambda + s\lambda\xi} d\lambda, \quad (3.6)$$

in which $d = 1/v$ is the slowness of the crack velocity and $\xi = x - vt$. The applied traction on crack faces as expressed in (3.6), has the functional form $e^{s\lambda\xi}$. Since the Laplace transform solutions of applying traction $e^{s\eta\xi}$ on crack faces have been solved in the previous section, the diffracted field generated from the propagating crack tip can be constructed by superimposing the fundamental solution and the stress distribution in (3.6). The result of shear stress expressed in the Laplace transform domain will be

$$\bar{\tau}_{yz}^d(\xi, y, s) = \frac{1}{2\pi i} \int_{\Gamma_{\eta_1}} \frac{pd}{(\eta_1 - d)} e^{sh(1-b_1v)\eta_1} \times \left\{ \frac{1}{2\pi i} \int_{\Gamma_{\eta_2}} \frac{\alpha_{1+}^*(\eta_2)}{(\eta_1 - \eta_2)\alpha_{1+}^*(\eta_1)} e^{-s[\alpha_1^* y - \eta_2 \xi]} d\eta_2 \right\} d\eta_1, \quad (3.7)$$

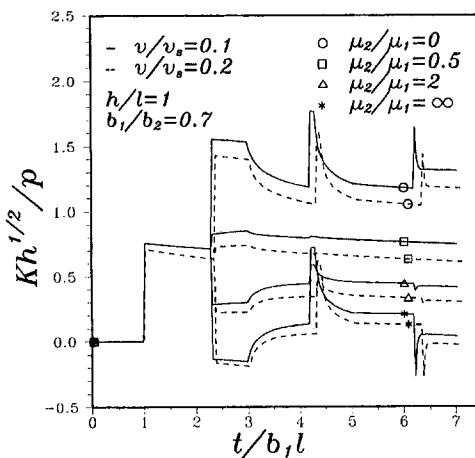


Fig. 5 Transient response of the dynamic stress intensity factor for different ratio of shear modulus of the layered medium

The exact transient solutions for a propagating crack in an unbounded medium in time domain can be obtained by inverting the Laplace transform domain of (3.7). The result is

$$\tau_{yz}^d(\xi, y, t) = \frac{-1}{2\pi^2} \int_{b_1 h}^{t-t_d} \operatorname{Re} \left[G(\eta_1^+, \eta_2^+) \frac{\partial \eta_1^+}{\partial t_1} \frac{\partial \eta_2^+}{\partial t_2} - G(\eta_1^-, \eta_2^+) \frac{\partial \eta_1^-}{\partial t_1} \frac{\partial \eta_2^+}{\partial t_2} \right] dt_1, \quad (3.8)$$

where

$$G(\eta_1, \eta_2) = \frac{pd\alpha_{1+}^*(\eta_2)}{(\eta_1 - d)\alpha_{1+}^*(\eta_1)(\eta_1 - \eta_2)},$$

$$\eta_1^\pm = \frac{-t_1}{h(1 - b_1 v)} \pm i\varepsilon,$$

$$\eta_2^\pm = \frac{-(\xi t_2 + b_1^2 v y^2) \pm iy \sqrt{t_2^2 - b_1^2 [y^2 + (\xi + vt_2)^2]}}{\xi^2 + (1 - b_1^2 v^2) y^2},$$

$$\frac{\partial \eta_2^\pm}{\partial t_2} = \frac{-\xi + iy [t_2^2 - b_1^2 [y^2 + (\xi + vt_2)^2]]^{-1/2} [t_2(1 - b_1^2 v^2) - b_1^2 v \xi]}{\xi^2 + (1 - b_1^2 v^2) y^2},$$

$$t_d = \frac{b_1 \{ b_1 v \xi + [\xi^2 + (1 - b_1^2 v^2) y^2]^{1/2} \}}{1 - b_1^2 v^2},$$

$$t_1 + t_2 = t.$$

The dynamic stress intensity factor for a propagating crack at an infinite medium can also be constructed by a similar manner. The result in the Laplace transform domain can be obtained from (2.8) and (3.6) and is expressed as follows:

$$K^d(s) = \frac{1}{2\pi i} \int_{\Gamma_\lambda} \frac{pd}{(\lambda - d)} e^{sh(1-b_1v)\lambda} \times \left\{ \frac{-\sqrt{2}\sqrt{1-b_1v}}{\sqrt{s}\alpha_{1+}^*(\lambda)} \right\} d\lambda. \quad (3.9)$$

The inversion Laplace transform of (3.9) will have the following form:

$$K^d(t) = p \sqrt{\frac{2}{\pi[v(t - b_1 h) + h]}} \times (1 - b_1 v)^{1/2} H(t - b_1 h). \quad (3.10)$$

The expression for $K^d(t)$ in (3.10) has the interesting form of the product of a function of the crack velocity $(1 - b_1 v)^{1/2}$ and the corresponding stationary crack solution $K^s(t)$ in (3.5) with a distance $v(t - b_1 h) + h$ from the crack tip. The value $(1 - b_1 v)^{1/2}$ is an universal function which depends only on crack speed and material properties. After some later time, the incident wave (i wave) generated from the dynamic concentrated loading and the diffracted wave (d wave) radiated out from the propagating crack will be reflected from the interface which will be indicated as the ir and dr wave, respectively. The solutions for reflected waves generated from the interface expressed in the Laplace transform domain can be obtained as follows:

$$\tau_{yz}^{ir+dr}(\xi, y, s) = \frac{1}{4\pi^2} \iint \frac{\gamma_{1/2}^*(\eta_2) pd\alpha_{1+}^*(\eta_2)}{(\eta_1 - d)(\eta_1 - \eta_2)\alpha_{1+}^*(\eta_1)} \times e^{sh(1-b_1v)\eta_1} e^{s\alpha_{1+}^*(y-2l) + s\eta_2^2} d\eta_2 d\eta_1, \quad (3.11)$$

where

$$\gamma_{1/2}^* = \frac{\mu_1 \alpha_1^* - \mu_2 \alpha_2^*}{\mu_1 \alpha_1^* + \mu_2 \alpha_2^*}.$$

The induced dynamic stress intensity factor by reflected ir and dr waves can be obtained by setting $y = 0$ in (3.11) and the fundamental solution expressed in (2.8). The result for the stress intensity factor expressed in the Laplace transform domain will have the following form:

$$\bar{K}^{ir+dr+d}(\lambda) = \frac{1}{4\pi^2} \iint \frac{pd\sqrt{2}\sqrt{1-b_1v}\gamma_{1/2}^*(\eta_2)}{\sqrt{s}(\eta_1 - d)(\eta_1 - \eta_2)\alpha_{1+}^*(\eta_1)} \times e^{sh(1-b_1v)\eta_1 - 2s\alpha_{1+}^*(\eta_2)l} d\eta_2 d\eta_1. \quad (3.12)$$

The dynamic stress intensity factor expressed in time domain will be

$$K^{ir+d}(t) = \frac{-\sqrt{2}pd\sqrt{1-b_1v}}{\pi^{3/2}} \int_{t_{ird}}^t \frac{1}{\sqrt{t-\tau}} \times \operatorname{Im} \left\{ \frac{\gamma_{1/2}^*(\lambda^+) \frac{\partial \lambda^+}{\partial t}}{(\lambda^+ - d)\alpha_{1+}^*(\lambda^+)} \right\}_{t=\tau} d\tau, \quad (3.13)$$

where

$$\lambda^\pm = \frac{-h(1 - b_1 v)t - 4b_1^2 v l^2 \pm i2l \sqrt{t^2 - b_1^2 \{ [vt + h(1 - b_1 v)]^2 + 4l^2 \}}}{h^2(1 - b_1 v)^2 + 4l^2(1 - b_1^2 v^2)},$$

$$t_{ird} = \frac{b_1^2 v h(1 - b_1 v) + b_1 \sqrt{h^2(1 - b_1 v)^2 + 4l^2(1 - b_1^2 v^2)}}{1 - b_1^2 v^2},$$

and

$$K^{dr+d}(t) = \frac{1}{2\pi^2} \int_{b_1 h + t_{drd}}^t \int_{b_1 h}^{t-t_{drd}} \frac{\sqrt{2(1-b_1v)}}{\sqrt{\pi(t-\tau)}} \times \operatorname{Re} \left[\frac{\gamma_{1/2}^*(\eta_2^+) G(\eta_1^+, \eta_2^+) \frac{\partial \eta_1^+}{\partial t_1} \frac{\partial \eta_2^+}{\partial t_2}}{\alpha_{1+}^*(\eta_2^+)} - \frac{\gamma_{1/2}^*(\eta_2^+) G(\eta_1^-, \eta_2^+) \frac{\partial \eta_1^-}{\partial t_1} \frac{\partial \eta_2^+}{\partial t_2}}{\alpha_{1+}^*(\eta_2^+)} \right]_{t=\tau} dt_1 d\tau, \quad (3.14)$$

where

$$\eta_1^\pm = \frac{-t_1}{h(1 - b_1 v)} \pm i\varepsilon,$$

$$\eta_2^\pm = \frac{-2b_1^2 v l \pm i \sqrt{(1 - b_1^2 v^2) t_2^2 - 4l^2 b_1^2}}{2l(1 - b_1^2 v^2)},$$

$$t_{drd} = \frac{-2lb_1}{\sqrt{(1 - b_1^2 v^2)}},$$

$$t_1 + t_2 = t.$$

Finally, the complete transient solutions of dynamic stress intensity factor that account for the contributions of all the reflected and diffracted waves are obtained explicitly. The solutions can be simplified into a very compact form as follows:

$$K(t) = K^0(t) + \sum_{m=1}^{\infty} 2K^m(t) + \sum_{n=1}^{\infty} 2^n K^{0,n}(t) + \sum_{m=1}^{\infty} \sum_{n=1}^{\infty} 2^{n+1} K^{m,n}(t), \quad (3.15)$$

where

$$K^0(t) = p \sqrt{\frac{2}{\pi[v(t - b_1 h) + h]}} \times (1 - b_1 v)^{1/2} H(t - b_1 h), \quad (3.16)$$

$$K^m(t) = \frac{p\sqrt{2(1-b_1v)}}{\pi^{3/2}} \int_{t_{v,m}}^t \frac{1}{\sqrt{t-\tau}} \times \text{Im} \left\{ \frac{[\gamma_{1/2}^*(\eta_{1,m}^+)]^m}{(1-\eta_{1,m}^+v)\alpha_{1+}^*(\eta_{1,m}^+)} \frac{\partial \eta_{1,m}^+}{\partial t_1} \right\}_{t_1=\tau} d\tau, \quad (3.17)$$

$$K^{m,n}(t) = \frac{2p\sqrt{2(1-b_1v)}i^q}{\sqrt{\pi}(2\pi i)^{n+1}} \int_{t_{v,m}+nt_{rr}}^t \int_{t_{v,m}}^{a_1} \int_{t_{rr}}^{a_2} \int_{t_{rr}}^{a_3} \dots \int_{t_{rr}}^{a_n} \frac{1}{\sqrt{t-\tau}} \text{SIFV}_{m,n} dt_n dt_{n-1} \dots dt_1 d\tau, \quad (3.18)$$

in which

$$a_1 = \tau - nt_{rr},$$

$$a_j = \tau - t_1 - t_2 - \dots - t_{j-1} - (n-j+1)t_{rr},$$

$$j = 2, 3, \dots, n$$

$$q = 0 \quad \text{when } n = 1, 3, 5, \dots;$$

$$q = 1 \quad \text{when } n = 2, 4, 6, \dots$$

$$\text{SIFV}_{m,n} = \text{Re} \left[\frac{[\gamma_{1/2}^*(\eta_{1,m}^+)]^m \gamma_{1/2}^*(\eta_2^\pm) \gamma_{1/2}^*(\eta_3^\pm) \dots \gamma_{1/2}^*(\eta_{n+1}^\pm) \left(\pm \frac{\partial \eta_{1,m}^+}{\partial t_1} \right) \dots \left(\pm \frac{\partial \eta_n^\pm}{\partial t_n} \right) \left(\pm \frac{\partial \eta_{n+1}^+}{\partial t_{n+1}} \right)}{(1-\eta_{1,m}^\pm v)\alpha_{1+}^*(\eta_{1,m}^\pm)(\eta_{1,m}^\pm - \eta_2^\pm)(\eta_2^\pm - \eta_3^\pm) \dots (\eta_{n-1}^\pm - \eta_n^\pm)(\eta_n^\pm - \eta_{n+1}^+)} \right]_{t=\tau} \quad \text{for } n = 1, 3, 5, \dots$$

$$\text{SIFV}_{m,n} = \text{Im} \left[\frac{[\gamma_{1/2}^*(\eta_{1,m}^\pm)]^m \gamma_{1/2}^*(\eta_2^\pm) \gamma_{1/2}^*(\eta_3^\pm) \dots \gamma_{1/2}^*(\eta_{n+1}^\pm) \left(\pm \frac{\partial \eta_{1,m}^+}{\partial t_1} \right) \dots \left(\pm \frac{\partial \eta_n^\pm}{\partial t_n} \right) \left(\pm \frac{\partial \eta_{n+1}^+}{\partial t_{n+1}} \right)}{(1-\eta_{1,m}^\pm v)\alpha_{1+}^*(\eta_{1,m}^\pm)(\eta_{1,m}^\pm - \eta_2^\pm)(\eta_2^\pm - \eta_3^\pm) \dots (\eta_{n-1}^\pm - \eta_n^\pm)(\eta_n^\pm - \eta_{n+1}^+)} \right]_{t=\tau} \quad \text{for } n = 2, 4, 6, \dots$$

$$\eta_{1,m}^\pm = \frac{-h(1-b_1v)t_1 - (2ml)^2 b_1^2 v \pm i2ml \sqrt{t_1^2 - b_1^2} \{ [vt_1 + h(1-b_1v)]^2 + (2ml)^2 \}}{h^2(1-b_1v)^2 + (2ml)^2(1-b_1^2v^2)}$$

$$\eta_j^\pm = \frac{-b_1^2 v l \pm i \sqrt{(1-b_1^2 v^2)t_j^2 - 4l^2 b_1^2}}{2l(1-b_1^2 v^2)}, \quad j = 2, 3, 4, \dots$$

$$t_{v,m} = \frac{b_1^2 v h(1-b_1v) + b_1 \sqrt{h^2(1-b_1v)^2 + (2ml)^2(1-b_1^2 v^2)}}{1-b_1^2 v^2},$$

$$t_j = t - t_1 - t_2 - \dots - t_{j-1}, \quad j = 2, 3, 4, \dots$$

$$\gamma_{1/2}^*(\eta) = \frac{\mu_1 \alpha_1^*(\eta) - \mu_2 \alpha_2^*(\eta)}{\mu_1 \alpha_1^*(\eta) + \mu_2 \alpha_2^*(\eta)},$$

$$\alpha_1^*(\eta) = \alpha_{1+}^*(\eta) \alpha_{1-}^*(\eta) = \sqrt{b_1 + \eta(1-b_1v)} \sqrt{b_1 - \eta(1+b_1v)},$$

$$\alpha_2^*(\eta) = \sqrt{b_2 + \eta(1-b_2v)} \sqrt{b_2 - \eta(1+b_2v)},$$

$$t_{rr} = \frac{2lb_1}{\sqrt{1-b_1^2 v^2}}.$$

For the numerical calculation of the transient stress intensity factor, we consider a semi-infinite crack in a layered medium and subjected to a symmetric dynamic concentrated loading on a stationary crack faces at time $t = 0$. At nondimensional delay time $t/b_1h = 1$, the crack starts to propagate with a constant speed v from the semi-infinite crack, the reflected and diffracted waves generated in the layered medium are shown in Fig. 2. The dynamic stress intensity factors of propagating crack for various situations are shown in Figs. 3–5. Figure 3 shows the dynamic stress intensity factor of a single strip for various crack

propagating speed. We can see that the stress intensity factor is large when the propagating speed is low. Figure 4 shows the dynamic stress intensity factor for different ratio of shear wave speed. It indicates that the stress intensity factor is small when the magnitude of the ratio b_1/b_2 is low. Figure 5 shows the transient dynamic stress intensity factor for different ratio of shear modulus of the layered medium, in which $\mu_2/\mu_1 = 0$ corresponds to the traction-free boundary condition and $\mu_2/\mu_1 = \infty$ corresponds to the rigid boundary condition. Figure 5 indicates that the traction-free boundary condition will cause the largest value of stress intensity factor in the transient analysis.

4 Conclusions

The phenomena of crack propagation, arrest, and branching are important subjects in the areas of dynamic fracture analysis. The interaction of reflected waves generated from boundaries with the moving crack had only been discussed in experimental works. Experimental results indicated that the reflected waves dominate the stability of crack propagation. It is very important to have the analytical results to investigate this important event. But it seems difficult to obtain the analytical solutions by using the well-known conventional method.

We have proposed a powerful superposition methodology and a useful fundamental solution is constructed in this study to solve the problem of a propagating crack in a layered medium. The fundamental solution is the problem of applying an exponentially distributed traction on the propagating crack face and the solution is determined by superposition of the fundamental solution in the Laplace transform domain. The dynamic crack propagation with constant velocity in a configuration of layered medium with a strip is investigated. An explicit and complete result of the dynamic stress intensity factor is obtained in closed form and numerical results are evaluated in detail. The numerical results show that the stress intensity factors induced by reflected waves from the interface are significant.

There still have many unanswered questions in dynamic fracture and this work may provide a useful technique for further investigation in more complicated dynamic fracture problems especially on the crack propagation event. The proposed method in this study has already been extended to solve more difficult in-plane problem of crack propagation with boundary effect, the results will be shown in a future paper.

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